NAME OF PILOT			TYPE OF	PLANE	ACCIDENT NUMBER			
GUNN, ROBERT	G		B-176		45-6-19-501			
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		Section G—POWE	R PLANT F	AILURE	\$3.5 25.5	at Light to	1781
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Section J—AIRPORT AND FACILITIES AND AIRWAYS	
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1. EXPLAIN NOME	
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Section K—WEATHER (This must be signed by weather officer of the reporting station)	er og skallen er
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2. IF WEATHER WAS A FACTOR IN THE ACCIDENT, STATE HOW AND ATTACH COPY OF WEATHER REPORTS	
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And brake resistance, orecly prior to loading.	
Tower is requesited Endia Arknowledgement from p lots on hydraulig t	Light of the
WEATHER OFFICER (Name, Grade, and Station)	
Section L—GENERAL INFORMATION	
1. IF ERROR ON THE PART OF SOMEONE OTHER THAN THE OPERATOR WAS A FACTOR, STATE HOW	
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NEO! MOTES	16-41009-1

1. Tell in Narrative Form, in as Much Detail as Necessary, Everthing That is Known about the Accident. Be Sure to Cover Everthing That May Have Contributed Toward the Accident. Include Recommendations for Action to Prevent Similar Accidents, and Action Taken

B-176 \$42-97280 was in a flight of aircraft being ferried from Great Ashfield to Valley for Redeployment to the Zone of Interior. The aircraft arrived over Valley at 1715 DEST and called Valley Tower for landing instructions. The pilot was informed by the tower that landing was to the North on Rumway One Four (14) (6000 Ft asphalt) and to stand by until called. The wind was N/NW at five (5) MPH. The Co-Pilot made his before landing check which includes checking the hydraulic pressure. The gauge indicated 650 lbs BSI. At 1730 DBST the tower gave the aircraft clearance to land. The pilot made a good landing 100 ft past the beginning of Runway one Four (14). One application of the brakes was made shortly after Linding. The brakes responded normally at this time. The aircraft was allowed to roll about one third the length of the runway and another application of the brakes was made. This time the brakes were completely inoperative, the aircraft started slowly off the runway to the right. The pilot tried to straighten the roll of the aircraft with throttle while the Co-Pilot attempted to build up hydraulic pressure with the emergency hand pump. This was done without success. The hydraulic pressure gauge indicated zero lbs PSI. With both brakes fully depressed the aircraft continued to roll on the grass area paralleling the runway. An attempt was made to ground loop the aircraft, but before it reacted to the power, the aircraft crossed the perimeter track at the end of the runway. At this point the mixtures and switches were cut; the aircraft rolled over a knell and dropped down into a hollow about six (6) ft deep. The left wheel broke completely off and the right landing gear collapsed causing the aircraft to come to rest in the hollow 100 yards past the end of the runway.

An investigation of the aircraft efter the accident revealed the following. Two of the batteries were checked and found to be serviceable with specific gravity readings of 1250, the other battery was damaged upon removal from the aircraft at the time of the accident, but from external appearence is presumed to have been serviceable. Examination of the fuse panel proved that neither the hydraulic switch fuse nor the hydraulic pump fuse was blown. The hydraulic system was operated and found to be functioning properly. The electric pump was fun, the hand pump operated and pressure and fluid obtained down to the brake deboesters on both main struts. The hydraulic system was intact as far as the brake deboesters after the accident. Fluid was obtained from the lines connecting from the deboesters to the wheels. The flexible hose connecting the deboester and wheel on the right gear was believed to have broken as a result of the wheel coming apart from the shock strut; hydraulic fluid was obtained from the ruptured hose line. The flexible hose line on the left wheel was damaged at the disconnect fitting. The brake expander tubes were not ruptured on either wheel.

Because the landing gear was so badly damaged as a result of the accident, the Aircraft Accident Investigating Committee was unable to determine whether or not there was a leak or a broken hydraulic line prior to the accident. Although this could not be definitely determined, it is the most probable explanation for the brake failure.

2. RECOMMENDATIONS

It is recommended that pilots be impressed with the importance of checking both the hydraulic pressure and actually depressing the brake pedals to check resistance prior to landing.

1-GENERAL INFORTAL BEING

1/-

& ACTION TAKEN

Tower is requesting Radio Acknowledgement from pilets on hydraulic pressure and brake resistance checks prior to landing.

ACCIDENT STATION.	INVESTIGATING BOARD PI	OM REPORTING	STATION:	STATION:				
NAME-PARK	IN X TRE	NAME-MEMS		NAME-INTER	LIGENCE OFFICER			
Capt.	ORGANIZATION 1407th AAF Base	Uni	ORGANIZATION	GRADE	ORGANIZATION			
Namy Man	1 K.	NAME-MEDIC	AL OFFICER	NAME -RECO	RDEA			
Cant.	OBGANISATION . 140/th AAF Base	GRADE Unit	ORGANIZATION COLUMN THE THE THE	CRADE	ORGANIZATION			

RESTRICTED

20 June 1945

Upon arriving at Valley Airdrome from Great Ashfield at 17:15 DBST, June 19, 1945. I contacted the tower and received instructions to circle the field and await further instruction. At 17:30 the tower called and gave me landing instructions. joined the traffic pattern on the east side of the field as directed and flew around the field to make our approach. We went through the check list preparitory to landing on the down wind leg. At this time the hydraulic pressure was satisfactory being about 700 pounds P.S.I.

We made a normal approach from about two miles from the field at 130 mph. This I.A. S. was used to insure a safe gliding speed of the A/C which was loaded so that it weighed approximately 52,800 pounds at time of landing. A normal landing was made at approximately 100 yds from the end of the runway in use. A calm wind existed at the time. After landing, I used the brakes to slow the A/C up at which time they worked satisfactorily. The co-pilot started through his after-landing procedure, opening the cowl flass. The cowl flaps on humber one and two engines opened but on number three and four they remained shut which we did not observe until later.

About this time the A/C started to go to the right; I applied the left brake and right throttle, getting a response from the engines but no reaction from the brake. The brake pedal was fully depressed. I called out to the co-pilot that "we have no brakes" and he immediately started using the hand pump while I kept both brakes applied. No results were obtained and the A/C continued to roll towards the end of the runway on the grass, having left the runway at approximately two thirds of the way down. We then tried to ground loop to the left but before the A/C reacted to the throttle we came to the perimeter track. At this point Capt. Schott, 550 Squadron 8-3 Officer who was riding as a passenger came up in the cockpit between the pilot and co-pilot and having observed the difficulty moved the mixture controls to the cut off position and I turned off the switches.

We ran off the end of the runway and dropped off in a large hole, just beyond and to the right of the runway going about 40 MPH.

Three members of the crew and passengers were injured. They were as follows: Capt. Dery, who was hit in the side and back with the left check gun. He was exrayed and found to be only bruised. Sgt. Walker who cut one of his fingers on his left hand slightly. This was bandaged. Sgt. Elliot who was thrown against bulkhead number six and injured his shoulder. He was excaved and found to have shoulder.

The A/C sustained the following damage: All four propellors bent. Right wheel broken off. Left landing gear strut collapsed. Flaps bent. Chin and ball buret broken. Plexiglass nose section broken. Right wing and under side of fuselage bent and scratched

ROBERT G. GUNN.

2nd IM., 0784146

RESTRICTED

STATEMENT

Coming into the field on the approach, I checked the hydrolic pressure and it read 650 lbs. Right after the landing when we had plane under control I went through my after landing check about the same stime I had finished, the plane started right off the runway, and the pilot told me the breaks were out. I looked at the pressure and the gage was at zero, and I immediately started to operate the hand pump but could nt build up any pressure.

About that time the pilot satrted to bring the airplane back on the runway, by then we had exhausted our runway, and continued on off the end.

Right after landing we had pressure to work breaks, and also to open two cowl flaps, but it went right down and couldn't build up any with band pump.

RICHARD O. ETTLING, Jr

2nd. Lt.



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RESTRICTED

STATEMENT

24 June 1945

Brake failure to B-17G 42-97280

Left outer expander tube changed before ship took off. System was bled and all connections checked for leaks by continual pressure of parking brakes for at least a day, and by appling added pressure on pedals. Hydraulic Reservoir filled.

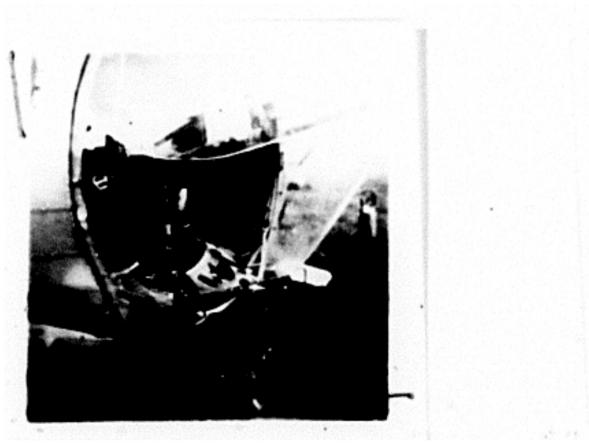
On taxiing out for take off brakes worked normally and hydraulic pressure was maintained by hydraulic motor until time of take-off.

On landing brakes held when pressure was first applied but failed to react again. The hydraulic hand pump was operated but was not necessary because motor was running at the time brakes failed. Expander tube was changed due to excessive time. All others changed at 1000 hrs.

M/Sgt. EDWARD W. HALLISEY Crew Chief. #280

m Soft Edward Halling

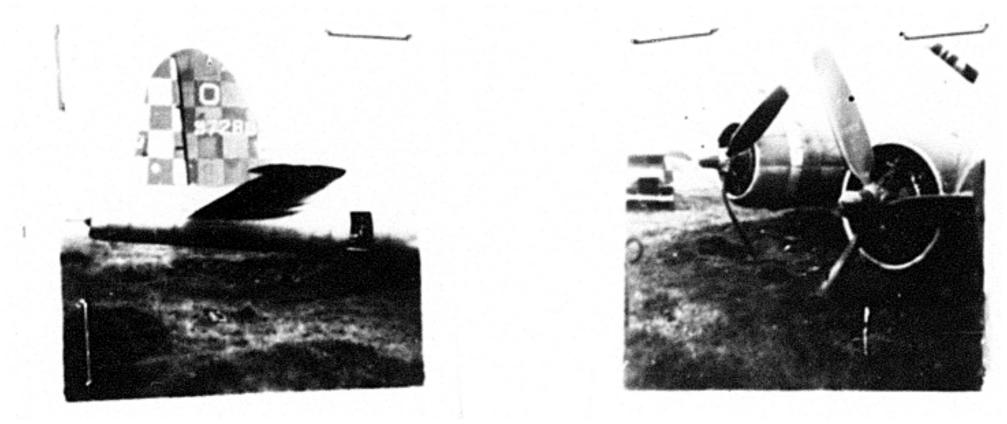


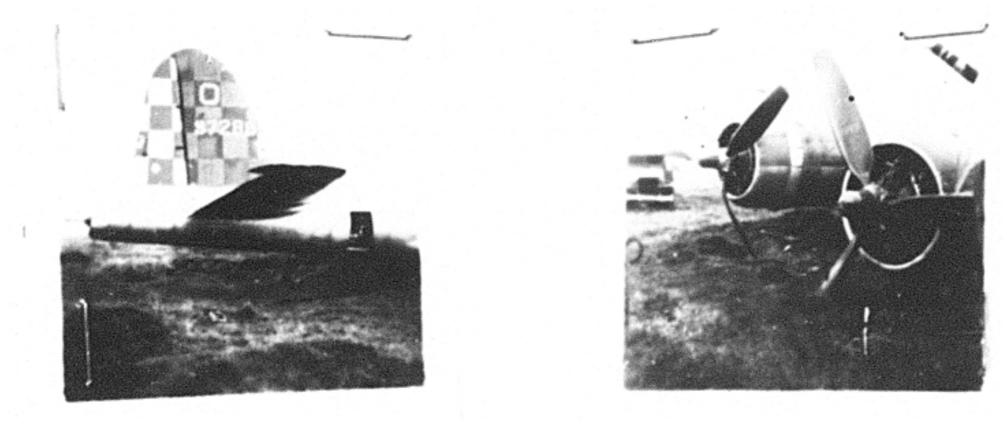


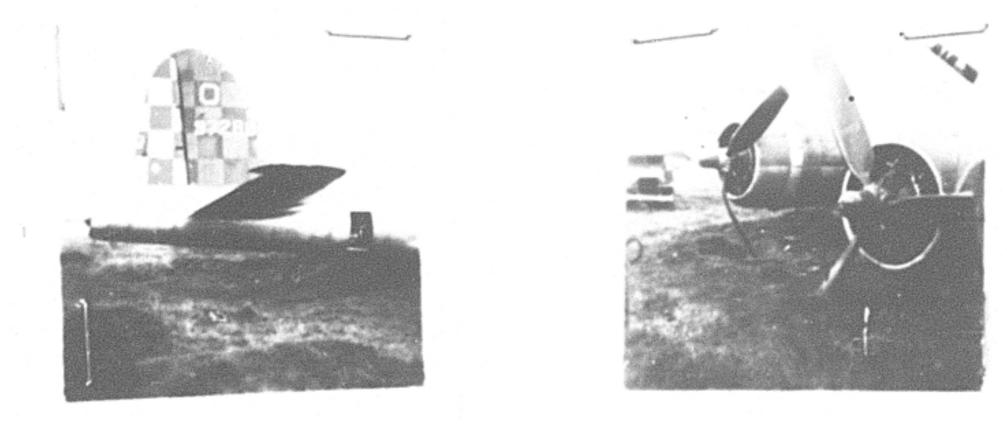












HEADQUARTERS

EURCPEAN DIVISION, AIR TRANSPORT COMMAND APO 741 C/p PM NEW YORK, HY

360.33

19 July 1945

SUBJECT: Transmittal of Report of Major Accident, AAF Form No. 14.

TO : Commanding General, AAF, Winston-Sahem, H.C. Attn: Chief, Flying Safety.

1. Transmitted herewith Repert of Major Accident, AAF Form No. 14 for B-17G 42-97280.

2. Delay of attached report, awaiting a maintenance check of landing goar of aircraft.

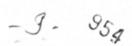
For the Commanding Officer:

SOYGE C. WEET, 1st Lt. Air Corps, Adjutant.

Incl: AAF Form No. 14. and 4 photes. with statements.

DISTRIBUTION:

Commanding General, AAF, Winston-Salem, N.C.
Attn: Chief, Flying Safety.
Commanding General, Eq. 5th Air Force,
AAF Sta. 101, APO 633, U.S. Army(thru channels)
Medical Officer, 1407th AAF Base Unit, European
Division, ATC, APO 741, U.S. Army.



AFAFS-1

BRIEF NO ...

DATE AND TIME

19 June 1945 17372

PLACE OF ACCIDENT

Valley, Wales

AIRCRAFT AND STATION

Probable major damage

AIRCRAFT DAMAGE

PILOT NAME

2nd Lt. Robert G. Gunn

PILOT STATION

PERSONNEL INVOLVED

8 AF, 550 Bomb, Sq., 385 Bomb, Grp., Ashfield, England

B-17 (42-97280) 385 Bomb, Gro., Great Achfield, England

Gunners: Sgt. Arnold L. Walker, Jr.; Set. Michael S. Elliett:

Passenger, Capt, William K. Dery, Jr. -- minor injury. Pilet:

AND INJURIES

Co-pilot, 2nd Lt. Richard O. Ettling: Havigator, 1st Lt.

Arthur F. Hymphrey, Jr.; Radio Operator, Sgt. Keith D. Stokes;

Gunner, Sgt. Horman O. Krone; Sgt. Charles J. Kaplin: Sgt.

Albert J. Casini: Passangers: M/Sgt. Edward W. Hallisey: Sgt.

R. L. RayMeans; Cpl. J.J. Redovicx; Capt. F.H. Schett; (ever)

DESCRIPTION OF ACCIDENT

Hydraulic system failed as brakes were synlied.

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950

SAMPL G

MISSION

Visibility 10 to 15 miles, wind MNV 3 MPH, 2/10 to 3/10 cloud

above 10,000 ft.

Angelsey Island, Wales DTG 20/00232 IH-18715 (20 June 45)

of Agg. Courtes T. Children's -- no injury.

5-5260, AF